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NUMERICAL STUDY ON THE WAKE EVOLUTION OF CONTRA-ROTATING PROPELLER IN PROPELLER OPEN WATER AND SELF-PROPULSION CONDITIONS

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Summary

In this study, the wake characteristics of a contra-rotating propeller (CRP) were investigated using a numerical simulation. The numerical simulation was carried out with a Reynolds averaged Navier–Stokes equations solver. The numerical simulations were performed on CRPs in both propeller open water and self-propulsion conditions to investigate their wake evolution characteristics. To study the effect of the rudder on the wake in the self-propulsion condition, the numerical simulations with and without a rudder were compared. The evolution of the CRP wake was analysed through velocity and vorticity contours on one transverse plane between the forward and aft propellers and two transverse planes located downstream of the CRP. The variations of thrust and torque of the forward and aft propellers during one revolution of the CRP were compared to investigate the interaction between forward and aft propellers and the effect of a rudder.

Key words: contra-rotating propeller (CRP); propeller wake; wake evolution; computational fluid dynamics (CFD); propeller-rudder interaction

1. Introduction

Reducing fuel consumption has always been a high priority for ship owners and ship operators. Therefore, shipyards are also making efforts to develop new devices to reduce hull resistance and improve propulsive efficiency. Recently, the regulations of International Maritime Organization (IMO) requiring the control of CO₂ emission with the Energy Efficiency Design Index (EEDI) have exaggerated the requirement of the development of energy saving devices. As a result, many types of energy saving devices have been suggested, and some were applied to real ships. There is no doubt that one of the most effective devices applied to real ships to improve the propulsive efficiency is the contra-rotating propeller (CRP), even though there are still some limitations in its application to all types of vessels because of large engine room space and high initial investment costs.

The fundamental approach to understanding the characteristics of propeller operation is the investigation of the evolution of propeller wake. The wake evolution was studied by an experimental measurement and a numerical simulation. The experimental measurement was performed with a pitot tube or a laser Doppler velocimetry (LDV) [1, 2], but most recent wake measurements were carried out using a particle image velocimetry (PIV). Tukker et al. [3] and Felli et al. [4] showed the availability of PIV to measure propeller wake. Di Felice et al. [5] examined the characteristics of propeller wake evolution at different loading conditions. Paik et al. [6] investigated the characteristics of the tip vortex and trailing vortex of a propeller. The mechanisms of the evolution of propeller wake in the transition and far fields were researched by Felli et al. [7]. The propeller wake field according to the variation of the propeller phase angle was studied by Hwang et al. [8]. Felli et al. [9] measured wake field and pressure fluctuation downstream of the propeller to investigate the correlation of the velocity and pressure signal. On the other hand, numerical simulations using computational fluid dynamics (CFD) based on Reynolds averaged Navier-Stokes (RANS) equations were performed by Rhee and Joshi [10], Di Felice et al. [11], Arikan et al. [12], Wang and Walters [13], Baek et al. [14], and Kinaci and Gokce [15] to study the characteristics of propeller wake.

Numerical methods based on potential theory have been applied to study the characteristics of CRP wake for a long time (Tsakonas et al. [16], Hoshino [17], and Paik et al. [18]). However, Paik et al. [19] applied CFD based on the RANS equations to investigate the characteristics of wake as well as shaft forces according to the combination of number of blades for the forward and aft propellers of CRPs. Paik et al. [20] performed PIV measurements and CFD simulations to survey the wake field of CRPs in detail.

In this research, a numerical investigation was performed to study the wake evolution of a CRP in propeller open water (POW) and self-propulsion conditions. The evolution of CRP wake was analysed through velocity and vorticity contours on one transverse plane between the forward and aft propellers and two transverse planes located downstream of the CRP. To study the wake effect of a rudder in the self-propulsion condition, the numerical simulations were compared with and without a rudder. And the variations of thrust and torque during one revolution were compared to investigate the interactions between the forward and aft propeller and between the CRP and the rudder.

2. Method of numerical simulation

2.1 Numerical models

A numerical simulation was performed using a commercial CFD code (FLUENT ver. 14) based on RANS equations. The governing equations for the conservations of mass and momentum are described as follows:

$$\frac{\partial \rho}{\partial t} + \nabla \cdot \left(\rho \vec{u}\right) = 0 \tag{1}$$

$$\frac{\partial}{\partial t}(\rho\vec{u}) + \nabla \cdot (\rho\vec{u}\vec{u}) = -\nabla p + \nabla \cdot \vec{\tau} + \rho\vec{g}$$
⁽²⁾

The stress tensor τ is given by

$$\bar{\tau} = \mu \left[\left(\nabla \vec{u} + \nabla \vec{u}^T \right) - \frac{2}{3} \nabla \cdot \vec{u} I \right]$$
(3)

where μ is the molecular viscosity and *I* is the unit vector.

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The numerical method applied for the turbulence model was the Reynolds stress model (RSM), and pressure and velocity were coupled with the SIMPLEC (Semi-Implicit Method for Pressure Linked Equations-Consistent) algorithm. The applied numerical methods were previously verified by Paik et al. [20].

2.2 Simulation setup

The main particulars of the object ship, a 70K Product Carrier, used in this study are summarized in Table 1. The scale ratio of the model ship is 32.7273. The number of blades is four for both propellers. The diameter of the aft propeller is 88.9 % of the forward propeller diameter. The CRP was designed to maximize propulsive efficiency using backward rake for forward propeller and forward rake for aft propeller, which was proposed by Inukai [21]. The main particulars of the CRP are summarized in Table 2.

Table 1 Main particulars of 70K PC

LBP (m)	219.0
Breadth (m)	32.2
Draught (m)	12.2

	Forward propeller	Aft propeller
Diameter (m)	7.2	6.4
Number of blades	4	4
P/D @ 0.7 r/R	0.739	0.882
C/D @ 0.7 r/R	0.210	0.204
Rotation direction	Counter-clockwise	Clockwise

Table 2 Main particulars of CRP

The grid system for the POW condition as shown in Figures 1 and 2 has two sliding blocks for the forward and after propellers. Pyramid cells were used on the surface of the propeller blade and the boundary of the sliding blocks, and tetrahedral cells were filled inside the sliding block. Except for the sliding blocks, structure girds were employed in the other domains.



Fig. 1 Computational domain and boundary conditions for POW condition



Fig. 2 Sliding blocks for forward and aft propellers

The computational domain and boundary conditions for the self-propulsion condition are illustrated in Figure 3. Two sliding blocks were applied to simulate the rotation of forward and aft propellers as shown Figure 4. Several multi-blocks were applied for the wake region of the CRP and the gap between the forward and aft propellers as shown in Figures 5.



Fig. 3 Computational domain and boundary conditions for self-propulsion condition



Fig. 4 Grid structure of sliding blocks for forward and aft propellers.

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Fig. 5 Multi-block structures for self-propulsion conditions with a rudder

The numbers of grids for the blocks of the forward and aft propellers are about 599K and 595K, respectively. The total number of grids for the POW and self-propulsion conditions are approximately 2M and 3M, respectively, as summarized in Table 3.

Table	3	Number	of	orid	cells	for	simulati	ons
Lanc	•	runnoer	or	Silu	cons	101	Simulan	ons

	POW	Self-propulsion w/o rudder	Self-propulsion w/ rudder
Block I	599K		
Block II	595K		
Others	634K	1,793K	1,961K
Total	1,798K	2,957K	3,125K

Initially, the forward and aft propellers were installed with a phase angle difference of 45° as depicted in Figure 6. The blades are numbered according to the rotating direction of the propeller. "F" in the name of the blade stands for the forward propeller, and "A" stands for the aft propeller. The definition of propeller blade angle (θ) follows the rotational direction of the forward propeller.



Fig. 6 Blade position and notation of forward and aft propellers (looking upstream)

The rotational speeds of the forward and aft propellers were identical. In the POW condition, the inflow velocity was 2.709 m/s and the rotational speed of the forward and aft propellers was 15.0 rps. The inflow velocity and the rotational speed of the CRP in the self-propulsion conditions were 1.349 m/s and 7.47 rps, respectively. The apparent advance coefficients (J) of the forward and aft propellers for the POW and self-propulsion conditions were 0.821 and 0.923, respectively. The operation condition was the self-propulsion point estimated from a model test in a towing tank. The propellers rotated with a constant time step corresponding to the rotation angle of 1.5° to obtain 240 data during each revolution for the both conditions. The wake field was investigated at three positions in the propeller downstream as defined in Figure 7. The reference position is the after perpendicular (AP), and the three positions are defined as positions A, B, and C, which are located at 146.6 mm, 103.0 mm, and 66.1 mm from the AP, respectively.



Fig. 7 Positions for wake field investigation

3. Results and discussion

3.1 Propeller open water condition

The iso-surfaces of axial vorticity in the POW condition are shown in Figure 8. For the tip vortex, the blue colour is the vortex of the forward propeller and the red colour is that of the aft propeller. For the root vortex, the blue colour is for the aft propeller are clearly observed. The interaction between the tip vortices of the forward and aft propellers is not strong because the diameter of the aft propeller is about 89% of the forward propeller. On the other hand, the root vortex of the aft propeller extends downstream. The shed vortex of the aft propeller is also observed, while that of the forward propeller is not observed.

The variations of thrust and torque of the forward and aft propellers are compared during one revolution in Figure 9. They have eight peaks, which is twice the number of blades, owing to the interaction of the two propellers. The mean thrust of the forward propeller is higher than that of the aft propeller because the CRP was designed to distribute 55% and 45% of total power for the forward and aft propellers in the self-propulsion condition, respectively. The amplitude of variation of the forward propeller is greater than that of the aft propeller, which is the same tendency as Paik et al. [19].



Fig. 8 Iso-surfaces for axial vorticity ($\omega_r = \pm 30$) at POW condition (left: oblique view, right: rear view)



Fig. 9 Comparison of thrust and torque variations of CRP at POW condition

The velocity contours and vectors are compared at positions A, B, and C in Figure 10. The dotted circle in the figure denotes the propeller disk of the forward propeller and the view looks upstream. The strong counter-clockwise rotational component in the time-average velocity vectors observed at position A is due to the forward propeller operation, whereas the rotational component at positions B and C is mitigated owing to the interaction with the aft propeller. This explains the principals of the reduction of the rotational energy loss in the CRP. Instantaneous velocity contours ($\theta = 0^\circ$) at position A shows the interaction with the aft propeller with four isolated velocity contours. These isolated velocity contours make the thrust and torque of the forward and aft propellers fluctuating with eight peaks as shown in Figure 9. The characteristics of the wake according to the position can be explained more clearly with instantaneous axial vorticity distribution. A propeller blade and its tip vortex (F_B1_TV, F_B4_TV) of the forward propeller at position A have a phase angle difference of about 45° owing to the effect of the propeller pitch. The tip vortices (F B1 TV, F B4 TV) of the forward propeller almost disappear at position B, whereas the tip vortices (A_B1_TV, A_B4_TV) of the aft propeller are observed. The shape of the shed vortex (A_B4_SV) connecting between the root vortex (A_B4_RV) and the tip vortex (A_B4_TV) of the aft propeller at position B is more noticeable than that of the forward propeller at position A. The axial velocity around the tip vortex (A_B1_TV, A_B4_TV) of the aft propeller at position B is accelerated at position C, whereas the tip vortex of the aft propeller at position B is weakened at position C. As shown in Figure 8, the root vortex (A_B1_RV, A_B4_RV) of the aft propeller at position C remains, while the root vortex (F B1 RV, F B4 RV) of the forward propeller disappears.



Fig. 10 Time-averaged velocity contours and vectors (top), instantaneous ($\theta = 0^\circ$) velocity contours and vectors (middle), and axial vorticity contours (bottom) at positions A, B, and C

3.2 Self-propulsion condition

The effect of a rudder on the wake evolution of the CRP was studied at self-propulsion condition. The iso-surface of the axial vorticity of the cases with and without a rudder are compared and shown in Figure 11. The root vortices in the case with a rudder is slightly larger due to the interaction with the rudder, and a strong negative vortex (WSV, wake shear vortex), generated by crossing flow of the rotational components in hull wake and propeller swirl, observed on the top of propeller disappears due to the rudder. The root vortex (red) of the forward propeller is biased to the starboard side because the loading of the forward propeller at the starboard side is greater than that of the port side due to the tangential component of hull wake. However, the root vortex (blue) of the aft propeller balances well since the rotational component of hull wake is mitigated due to the forward propeller. Nevertheless, there is no significant difference in the vortex structures around the propellers for the both cases.

The thrust and torque in the case with a rudder are greater compared to that of the case without a rudder as illustrated in Figure 12. The increase of thrust and torque is due to the reflection effect of the rudder. The thrust increments of the forward and aft propellers due to the rudder are about 10.3% and 3.5%, respectively. And the torque increments of the forward

and aft propellers are about 6.8% and 4.7%, respectively. As a result, the propulsive efficiency of the forward propeller is increased, whereas that of the aft propeller is decreased.



Fig. 11 Comparison of iso-surface for axial vorticity between without (top) and with (bottom) rudder for self-propulsion condition ($\omega_x = \pm 30$)



Fig. 12 Comparison of thrust (top) and torque (bottom)

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The FFT analysis results of the thrust and torque are compared in Figure 13 for the cases with and without a rudder. In general, the amplitudes in even blade frequencies are greater than those in odd blade frequencies. Strasberg & Breslin [22] suggested Equations 4 and 5 to find the dominant frequencies of the axial and transversal fluctuating forces due to the interaction between the forward and aft propellers.

$$f = mZ_f N_f + nZ_a N_a \tag{4}$$

$$f' = m'Z_f N_f + n'Z_a N_a \tag{5}$$

where f and f' are the dominant frequencies for the axial and transversal forces, respectively. Z and N are the number of blades and the rotational speed of a propeller. Subscript f and a denote the forward and aft propellers, respectively. m and n are the natural numbers to satisfy the relation of $mZ_f = nZ_a$, and m' and n' are the natural numbers to satisfy the relation of $mZ_f \pm 1 = n'Z_a$. In other words, the 2nd and 4th order blade frequencies correspond to m = n = 1 and m = n = 2, and they are the dominant frequencies of thrust and torque variations. On the other hand, the 1st order blade frequency reflects the effect of hull wake rather than the transversal forces. The amplitude of the 1st order blade frequency for the case with a rudder increases slightly owing to the effect of the rudder, whereas the changes in the other frequencies are not distinct.



Fig. 13 FFT results of thrust (top) and torque (bottom) for the cases with and without rudder



Fig. 14 Time-averaged velocity contours and vectors (top), instantaneous ($\theta = 0^\circ$) velocity contours and vectors (middle), and axial vorticity contours (bottom) at positions A, B, and C for without-a-rudder condition

The velocity contours and vectors at positions A, B, and C for the cases without a rudder are illustrated in Figure 14. The axial velocity of the port side at position A is more accelerated due to the rotational direction of the forward propeller operating in the upward tangential inflow velocity, while the axial velocity of the starboard side at position B is accelerated more due to the rotational direction of the aft propeller operating in the swirling wake of the forward propeller. The axial velocity at position B is more accelerated overall at position C. In the vorticity contours, a pair of bilge vortex (port/starboard bilge vortex, PBV/SBV) generated from the bilge of hull are observed at the port and starboard sides of position A. The WSV is observed at the top of propeller disk at position A, and it moves to the starboard side at positions B and C. The tip vortices (F_B3_TV, F_B4_TV) of the forward propeller in the port side are stronger than the others (F_B1_TV, F_B2_TV) in the starboard side and the tip vortices (F_B1_TV, F_B4_TV) in the top side is stronger than the others (F_B2_TV, F_B3_TV) in the bottom side due to the axial and tangential components of the hull wake. On the other hand, the



Fig. 15 Time-averaged velocity contours and vectors (top), instantaneous ($\theta = 0^\circ$) velocity contours and vectors (middle), and axial vorticity contours (bottom) at positions A, B, and C for with-a-rudder condition

tip vortex (A_B4_TV) of the aft propeller at the top and port side is stronger than the others due to the rotating swirl of the forward propeller. All vortices of the forward and aft propellers are mitigated at position C.

The velocity contours and vectors at positions A, B, and C for the cases with a rudder are illustrated in Figure 15. The wake fields at positions A and B of Figure 14 and Figure 15 are very similar. However, the axial velocity at position C is almost symmetrical due to the rudder, and the WSV at the propeller top position moves to the starboard side due to the disturbance of the rudder. The tip vortex (A_B3_TV) of the aft propeller at position C is stronger than that of the case without a rudder.

4. Conclusions

In this research, the characteristics of wake evolution for CRP were studied using numerical simulations. The numerical simulations were performed in the propeller open water and self-propulsion conditions. In the self-propulsion conditions, the effect of a rudder was investigated through the cases with and without a rudder.

Flow structure at three transverse planes in propeller downstream were compared with velocity contours and vectors as well as axial vorticity contours. From the propeller open water condition, the reduction of tangential energy loss in CRP was investigated through the interaction of the forward and aft propellers. The interaction between the tip vortices of the forward and aft propellers is not strong because the diameter of the aft propeller is about 89% of the forward propeller, but the root vortex of the forward propeller disappears owing to the interaction with the aft propeller.

From the self-propulsion condition, the existence of rudder was not significantly affected on the wake field. However, there was obvious effect on increase of thrust and torque due to a rudder. As the effects of the rudder, the axial velocity around the leading edge of rudder was almost symmetrical and the WSV at the propeller top position moved to the starboard side.

The increments of thrust of the forward and aft propellers due to the rudder are about 10.3% and 3.5%, respectively. And the increments of torque of the forward and aft propellers are about 6.8% and 4.7%, respectively. As a result, the propulsive efficiency of the forward propeller is increased, whereas that of the aft propeller is decreased. The variation amplitude of thrust and torque of the forward propeller is greater than that of the aft propeller. The dominant frequencies of the variation are the 2^{nd} and 4^{th} order blade frequencies. The amplitude of the 1^{st} order blade frequency for the case with a rudder increases slightly owing to the effect of the rudder.

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